

# Adaptive Control of Time-Varying Parameter Systems



**Submitted for publication**

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- Existing literature only obtains **UUB** stability for slowly-varying parametric uncertainty
- Difficult to achieve **asymptotic tracking** because the time-derivative of parameter acts like an unknown exogenous disturbance in the parameter estimation dynamics
- Proposed method utilizes a **RISE-like update law** which compensates for potentially destabilizing terms arising due to time-varying nature of parameters
- **Asymptotic tracking result is achieved** via a Lyapunov-based design/analysis methods

- Consider a control-affine system

$$\dot{x}(t) = h(x(t), t) + d(t) + u(t)$$

- Function  $h(x, t)$  can be linearly parameterized as

$$h(x(t), t) \triangleq Y_h(x(t), t)\theta_f(t)$$

- Treating  $d(t)$  as a parameter, the system can be re-parameterized as

$$\dot{x}(t) = Y(x(t), t)\theta(t) + u(t)$$

where  $\theta(t) \triangleq \begin{bmatrix} \theta_f(t) \\ d(t) \end{bmatrix}$  and  $Y(x(t), t) \triangleq \begin{bmatrix} Y_h(x(t), t) & I_n \end{bmatrix}$



- Tracking error  $e \triangleq x - x_d \rightarrow 0$ , where

- Assumption 1

$$\|\theta(t)\| \leq \bar{\theta}, \|\dot{\theta}(t)\| \leq \zeta_1, \|\ddot{\theta}(t)\| \leq \zeta_2$$

- Assumption 2

$$\|x_d(t)\| \leq \bar{x}_d, \|\dot{x}_d(t)\| \leq \delta_1, \|\ddot{x}_d(t)\| \leq \delta_2$$

- Define filtered tracking error

$$r \triangleq \dot{e} + \alpha e$$

which yields

$$r = Y\theta + u - \dot{x}_d + \alpha e$$



- **Control input** is designed as

$$u \triangleq -Y_d \hat{\theta} - \alpha e + \dot{x}_d + \mu$$

which yields

$$r = Y\theta - Y_d \hat{\theta} + \mu$$

Taking time derivative yields

$$\dot{r} = (\dot{Y} - \dot{Y}_d)\theta + (Y - Y_d)\dot{\theta} + \dot{Y}_d \tilde{\theta} + Y_d \dot{\theta} - Y_d \dot{\hat{\theta}} + \dot{\mu}$$

- The **update law** is designed as

$$\dot{\hat{\theta}} \triangleq \text{proj}(\Lambda_0(t)) = \begin{cases} \Lambda_0, & \|\hat{\theta}\| < \bar{\theta} \vee (\nabla f(\hat{\theta}))^T \Lambda \leq 0 \\ \Lambda_1, & \|\hat{\theta}\| \geq \bar{\theta} \wedge (\nabla f(\hat{\theta}))^T \Lambda > 0, \end{cases}$$

where  $\Lambda_0 \triangleq \Gamma Y_d^T (Y_d \Gamma Y_d^T)^{-1} [\beta \text{sgn}(e)]$

$$\Lambda_1 \triangleq \left( I_{m+n} - \frac{(\nabla f(\hat{\theta}))(\nabla f(\hat{\theta}))^T}{\|\nabla f(\hat{\theta})\|^2} \right) \Lambda_0$$

where  $f$  is a continuously differentiable convex function

- The continuous auxiliary term  $\mu$  acts as a stabilizing term to cancel the side-effects of projection, and is designed as a solution to

$$\dot{\mu} \triangleq \begin{cases} \mu_0, & \|\hat{\theta}\| < \bar{\theta} \vee (\nabla f(\hat{\theta}))^T \Lambda \leq 0, \\ \mu_1, & \|\hat{\theta}\| \geq \bar{\theta} \wedge (\nabla f(\hat{\theta}))^T \Lambda > 0 \end{cases}$$

where

$$\begin{aligned} \mu_0 &\triangleq -Kr \\ \mu_1 &\triangleq \mu_0 - Y_d(\Lambda_0 - \Lambda_1) \end{aligned}$$

The closed loop dynamics for both cases

$$\dot{r} = (\dot{Y} - \dot{Y}_d)\theta + (Y - Y_d)\dot{\theta} + Y_d\tilde{\theta} + Y_d\dot{\theta} - \beta \operatorname{sgn}(e) - Kr$$

- Theorem 1. The designed controller and adaptation law ensure that the tracking error  $\|e(t)\| \rightarrow 0$  as  $t \rightarrow \infty$ , provided that the gain condition

$$\beta > \gamma_1 + \frac{\gamma_2}{\alpha}$$

is satisfied.

- Proof : Consider the candidate Lyapunov function

$$V_L(y(t), t) \triangleq \frac{1}{2}r^T r + \frac{1}{2}e^T e + P$$

where  $y(t) \triangleq \left[ z^T(t) \quad \sqrt{P(t)} \right]^T$



where  $P(t)$  is a generalized solution to

$$\dot{P}(t) \triangleq -L(t) \quad L \triangleq r^T (N_B - \beta \text{sgn}(e))$$

For the closed-loop error system, the Lyapunov derivative

$$\begin{aligned} \dot{\tilde{V}}_L &\stackrel{a.e.}{\subset} r^T (\tilde{N} + N_B - \beta \text{sgn}(e) - Kr - e) \\ &+ e^T (r - \alpha e) - r^T (N_B - \beta \text{sgn}(e)) \end{aligned}$$

Using **LaSalle-Yoshizawa Theorem for non-smooth systems (RT1)** yields semi-global asymptotic stability.

Future work:

Improve parameter estimation  
extension to NN and system identification

# Sparse Learning-Based Approximate Dynamic Programming with Barrier Constraints



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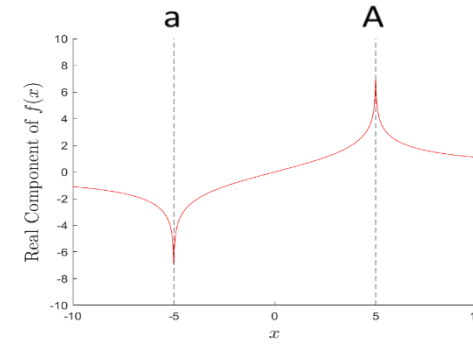
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# Barrier Function Questions

- Use model-based **RL (Actor-Critic (AC) or AC-Identifier)** to develop approximate solution to HJB equation (approximate optimal control)
- Advances in **Bellman Error (BE)** extrapolation for simulation of experience to achieve **Exploration AND Exploitation** for very fast learning
- **Sparse learning** allows for mixed density of basis functions and eliminates relearning of entire set of weights
- **Barrier functions** are known to provide state constraints for “safety”
  - How is BE extrapolation performed? How are off-trajectory points selected? Which states need to be transformed? Are points defined pre- or post- barrier state transform? Can sparse Bellman error extrapolation be used? Switching extrapolation stacks?
- **Open problems**
  - Since the barrier function makes the dynamics more complex, is the rate of learning affected? How does sparsity affect the computational power required? Zeroing Barrier Functions?

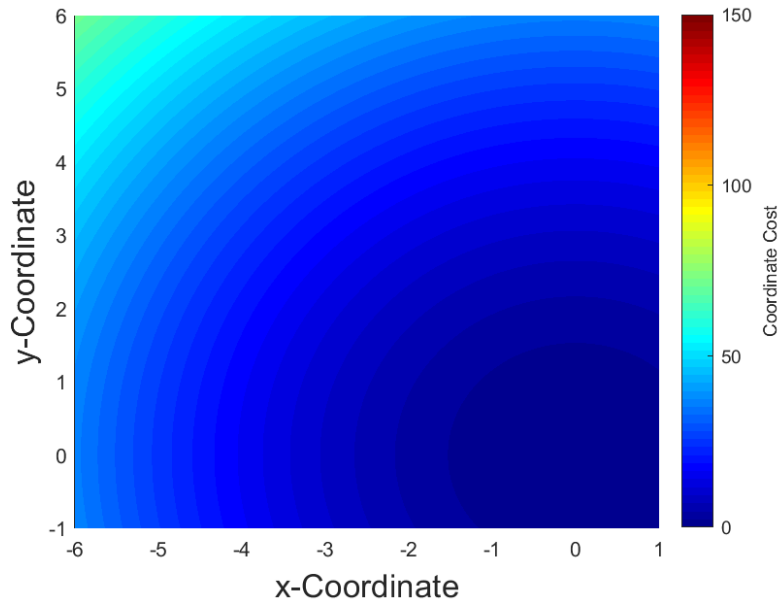
- Logarithmic Barrier Function

$$s = f(x, a, A) = \ln\left(\frac{A}{a} \frac{a-x}{A-z}\right)$$



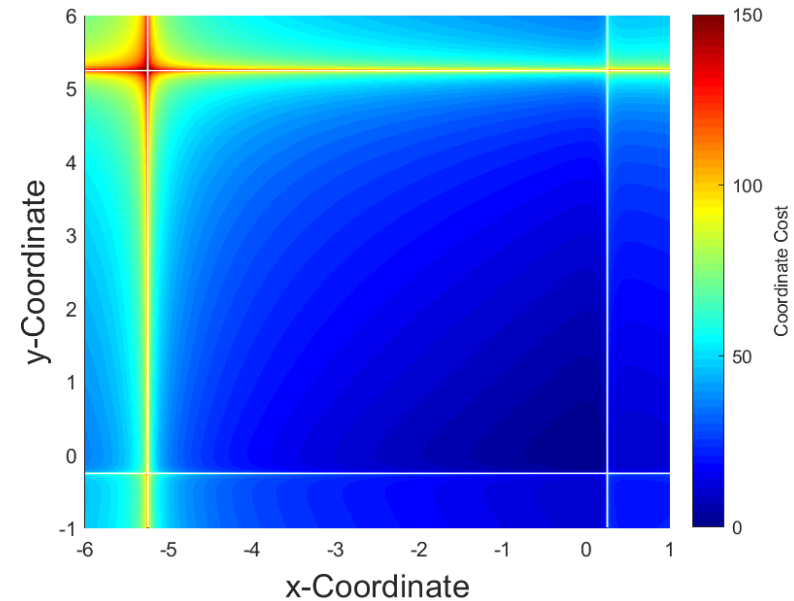
- Cost Before Barrier Function

$$xQx$$



- Cost After Barrier Function

$$sQs$$

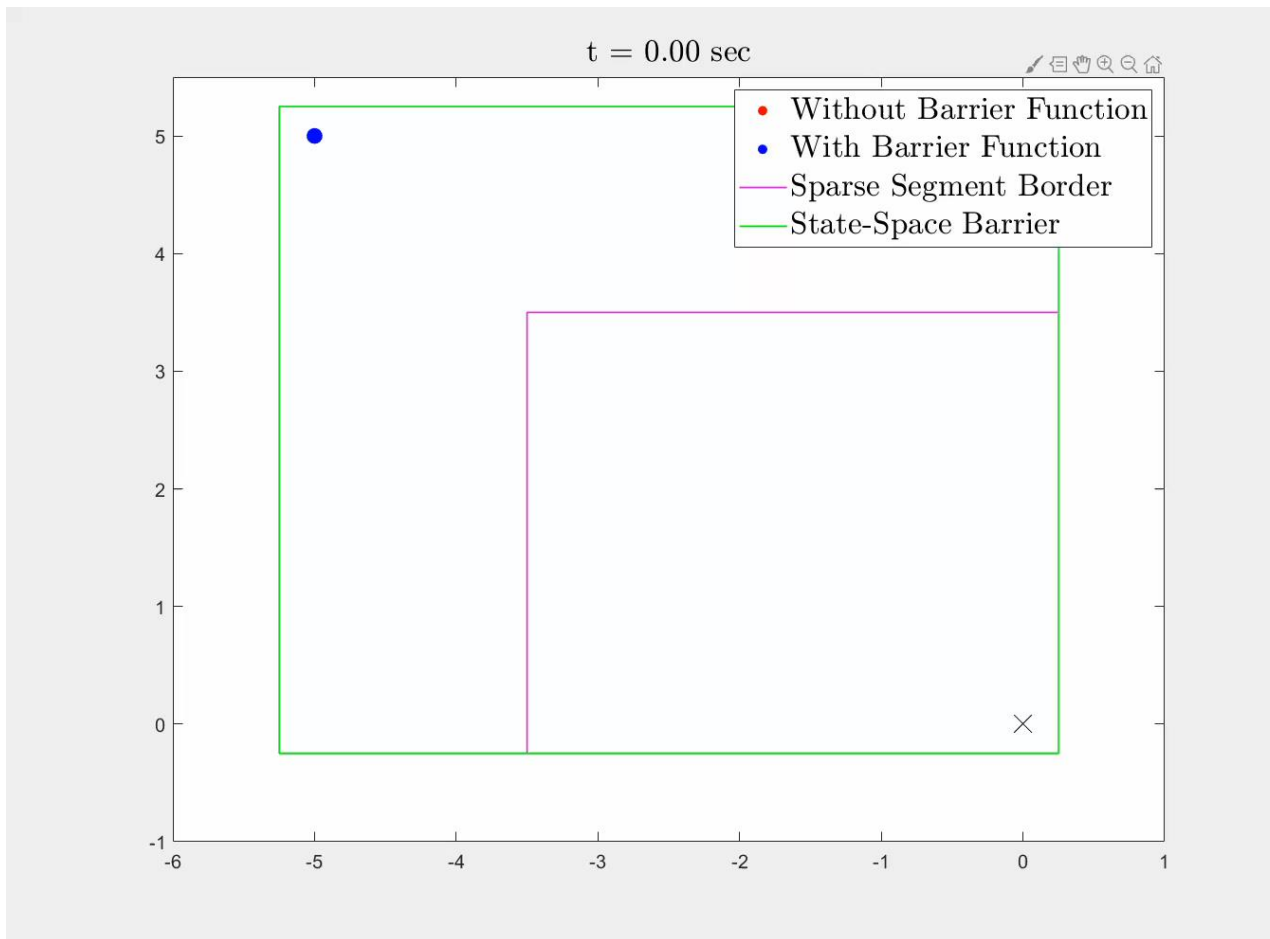




**Theorem** Given  $x(t)|_{t=0} \in (a, A)$ , using the class of dynamics  $\dot{s} = F(s) + G(s)u$ , and provided a sufficient number of BE extrapolation points are chosen and gains are selected according to sufficient conditions, then the system state  $s(t)$ , weight estimation errors  $\tilde{W}_c(t)$  and  $\tilde{W}_a(t)$ , and policy  $u(t)$  are uniformly ultimately bounded.

- From  $x = f^{-1}(a, A, s)$ ,  $x$  converges to a neighborhood of the origin, and hence, **the optimal policy is approximated.**

# Simulation Video



# Model-based Reinforcement Learning for Optimal Feedback Control of Switched Systems



Submitted for publication

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# Switched System ADP



- F-16 longitudinal dynamics
  - [Stevens, Lewis, Johnson, 2016]
- $x_1$  is Angle of Attack
- $x_2$  is Pitch
- $x_3$  is Pitch Rate
- $u$  is the change in thrust around the linearized point
- The control objective is to get to level flight ( $x_2=0$ )



	Dynamic Model
Mode 1, Unaltered Model	$\dot{x} = \begin{bmatrix} -1 & 0.9 & -0.002 \\ 0.8 & -1.1 & -0.2 \\ 0 & 0 & -1 \end{bmatrix} x + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u$
Mode 2, Altered Model	$\dot{x} = \begin{bmatrix} -0.8 & 0.2 & -0.01 \\ 0.6 & -1.3 & -0.1 \\ 0 & 0 & -1 \end{bmatrix} x + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u$
Mode 3, Altered Model	$\dot{x} = \begin{bmatrix} -1 & 0.5 & -0.02 \\ 0.9 & -0.8 & -0.4 \\ 0 & 0 & -1 \end{bmatrix} x + \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix} u$

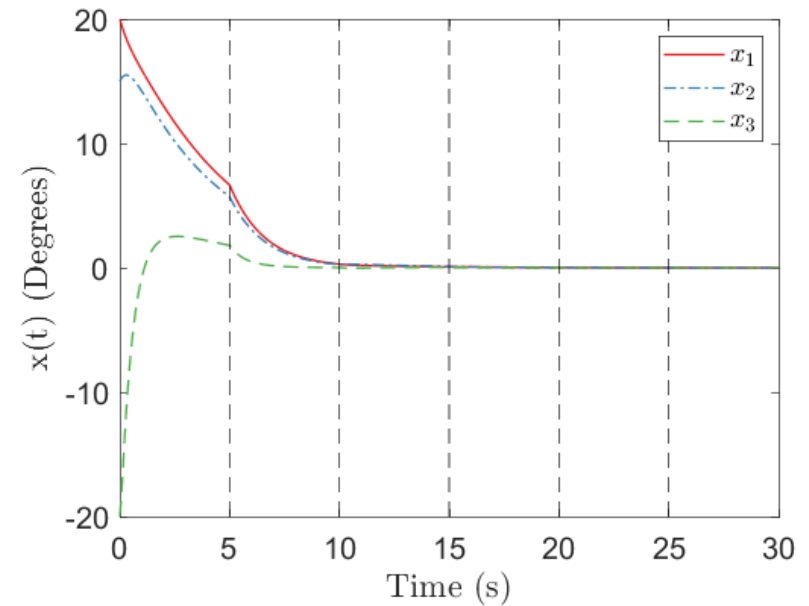
Parameter	Mode 1	Mode 2	Mode 3
Q	diag([1, 1, 1])	diag([5, 5, 5])	diag([3, 3, 3])
R	0.5	2	1
$\bar{\Gamma}$	$10^3$	$10^3$	$10^3$
$\underline{\Gamma}$	500	500	50
$\lambda$	0.4	0.5	0.5
$\nu$	0.005	0.005	0.005
$\eta_{c1}$	3	1	1
$\eta_{c2}$	5	2.5	5
$\eta_{a1}$	20	10	5
$\eta_{a2}$	1	0.75	1
N	10	10	10





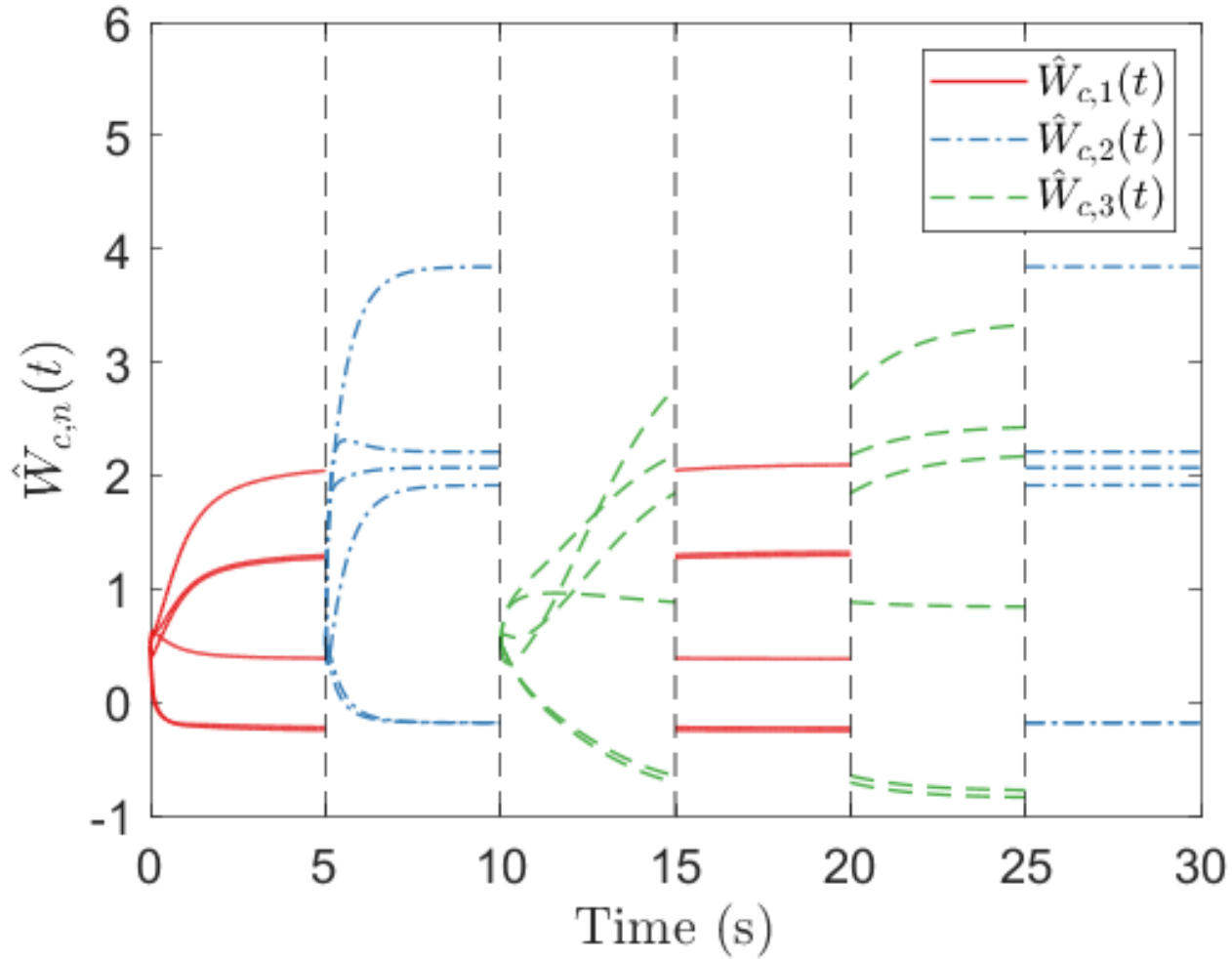
# Switched System ADP

- Switched System ADP
  - Preliminary Simulation Results
- Switch between multiple dynamical systems
  - Arbitrary switching sequence
  - Satisfies minimum dwell-time condition
- Switching Sequence
  - {1,2,3,1,3,2}





# Switched System ADP



# Switched System ADP

